In the Claims:

- 1. (Original) Gas turbine component, namely static gas turbine
 2 component, especially for an aircraft engine, characterized
 3 in that the same is formed at least region-wise of metal
 4 foam.
- 2. (Original) Gas turbine component according to claim 1, characterized in that the same forms a housing structure (19, 24, 26).
- Original) Gas turbine component according to claim 2, characterized in that, the housing structure (19, 24, 26) is formed at least region-wise or section-wise of metal foam (20), whereby the section or region of the housing structure formed of metal foam is connected with at least one section or region embodied as a carrier.
- 4. (Original) Gas turbine component according to claim 3, characterized in that the section or region formed of metal foam (20) is positioned in a sandwich-like manner between two plate-shaped carriers (21, 25) and connected therewith.

- 5. (Original) Gas turbine component according to claim 1, characterized in that the same forms an intake lining (27, 31, 32, 34).
- 6. (Original) Gas turbine component according to claim 5, characterized in that the intake lining (27, 31, 32, 34) of metal foam (28) is fixedly or rigidly connected with at least one carrier (29).
- 7. (Original) Gas turbine component according to claim 6, characterized in that the or each carrier (29) comprises openings or bored holes (35), and that the metal foam (20) of the intake lining (34) is embodied open-pored, so that both the or each carrier as well as the metal foam can be flowed-through for the cooling thereof.
- 1 8. (Original) Gas turbine component according to claim 1,
 2 characterized in that the same forms a pipe line system
 3 (38, 41, 42, 44, 47).
- 9. (Original) Gas turbine component according to claim 8, characterized in that the pipe line system (38, 41, 42, 44, 47) is formed by at least one closed-walled pipe (39), whereby the or each pipe (39) is surrounded by metal foam (40).

Claims 10 to 12 (Canceled).